

GOES-R Near Surface UAS Feasibility Demonstration Study

Supports: NESDIS Post-Launch Satellite Sensor Product Calibration & Validation
Funded by: GOES-R Program Office

Supporting Publications:

Francis Padula; Aaron J. Pearlman; Changyong Cao; Steve Goodman, "Towards post-launch validation of GOES-R ABI SI traceability with high-altitude aircraft, small near surface UAS, and satellite reference measurements," Proc. SPIE 9972, Earth Observing Systems XXI, 99720V (19 September 2016). ([link](#))

Aaron J. Pearlman; Francis Padula; Xi Shao; Changyong Cao; Steven J. Goodman, "Initial design and performance of the near surface unmanned aircraft system sensor suite in support of the GOES-R field campaign," Proc. SPIE 9972, Earth Observing Systems XXI, 99720U (19 September 2016). ([link](#))

Francis Padula, Aaron Pearlman, Xi
Shao, Changyong Cao, Steve Goodman

NOAA NESDIS

March 2017



GOES-R Field Campaign Overview

The purpose of the GOES-R field campaign is to support post-launch validation of L1b & L2+ products:

- **Advanced Baseline Imager (ABI) & Geostationary Lightning Mapper (GLM):**

- Planning ~6 week field campaign (~100 flight hours) with the high-altitude NASA ER-2 platform coordinated with ground based and near surface observations over several Earth targets
- Time-Frame :
 - March – May 2017
- An open access data portal will provide all validation datasets acquired during the campaign



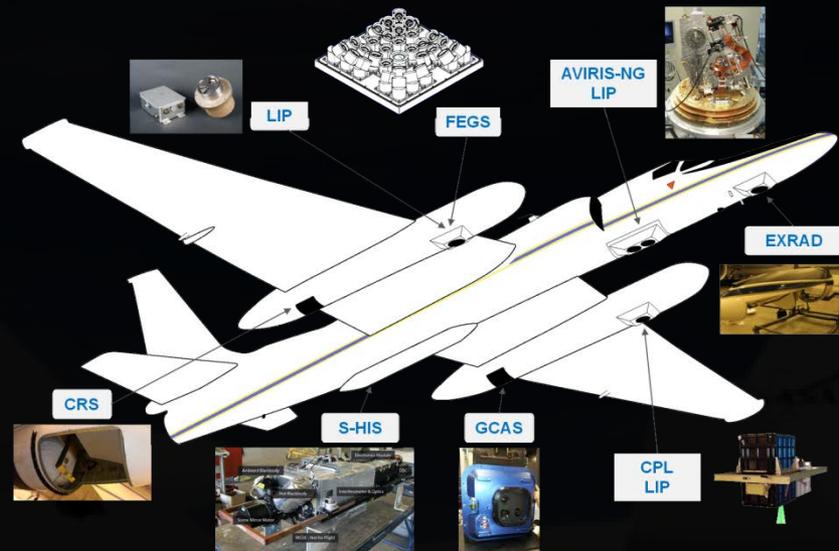
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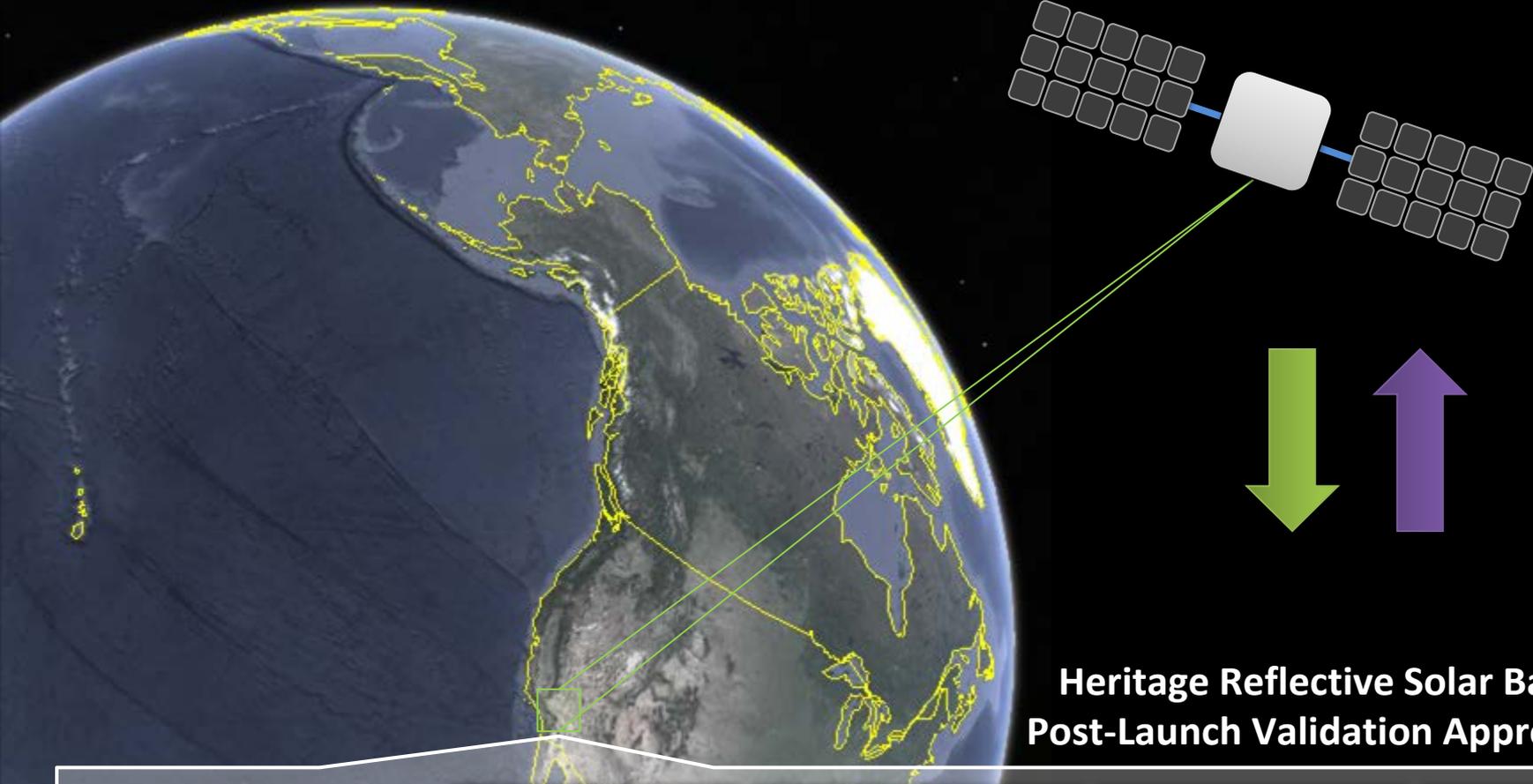
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GOES-R Field Campaign ER-2 Based Instruments



High to Moderate Resolution Satellite Sensors Leverage Small Uniform Earth Targets for Post-Launch Validation



Heritage Reflective Solar Band Post-Launch Validation Approach:

Collect high quality surface reference data that is directly compared to satellite observations



Post-Launch Validation Challenges & Gaps for Low Resolution Environmental Satellite Sensors

- » **Challenging to provide high quality data that can be directly compared to satellite observations without gross assumptions (i.e. uniformity):**
 - Ground validation measurements are typically point-based measurements
 - Often need to disturb the collection environment to make the measurements
 - Labor intensive
 - Costly (typically involves a large team)
 - Repeatability can be challenging
 - Limited collection geometry
- » **Currently no operational capability to measure goniometric observations over regions comparable to environmental satellite observations**
- » **Difficult to collect observations of extended regions**





Development of Advanced Post-Launch Validation Capabilities: Near Surface UAS Measurements



GOES-R Funded: "GOES-R Near Surface UAS Feasibility Demonstration Study" - NOAA Cooperative Institute Partnership with the University of Maryland (UMD) in collaboration with the NOAA UAS Program

Scope: Develop prototype UAS & assess the feasibility of near surface validation reference measurement capabilities in support of GOES-R Field Campaign validation efforts (L1b/L2+)

Phase 1: Procurement/Development & Integration of Prototype Systems:



Rotary UAS



Fixed-Wing UAS



Phoenix ACE XL Specifications
Endurance: 30 minutes
Fully autonomous system
Take-off weight: 10 lbs.

Customized electronic enclosure and autonomous 2 axis gimbal

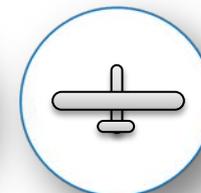
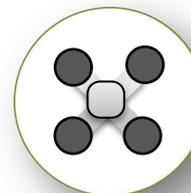


Talon120 Specifications
Length: 6' Wingspan: 12.5' MGTOW: 20 lbs.
Payload capacity: 2.5 lbs.
Range: 8 miles LOS
Endurance: 2.0 - 2.5 hours
Fully autonomous system
Typical operating alt.: 50-500 ft. AGL; MSL up to 10,000 ft

Customized nose cone for high resolution georeferenced imagery

Collection Reference Data:

- 1) **Rotary UAS** - Goniometric observations & area collection
- 2) **Fixed-Wing UAS** - area collection

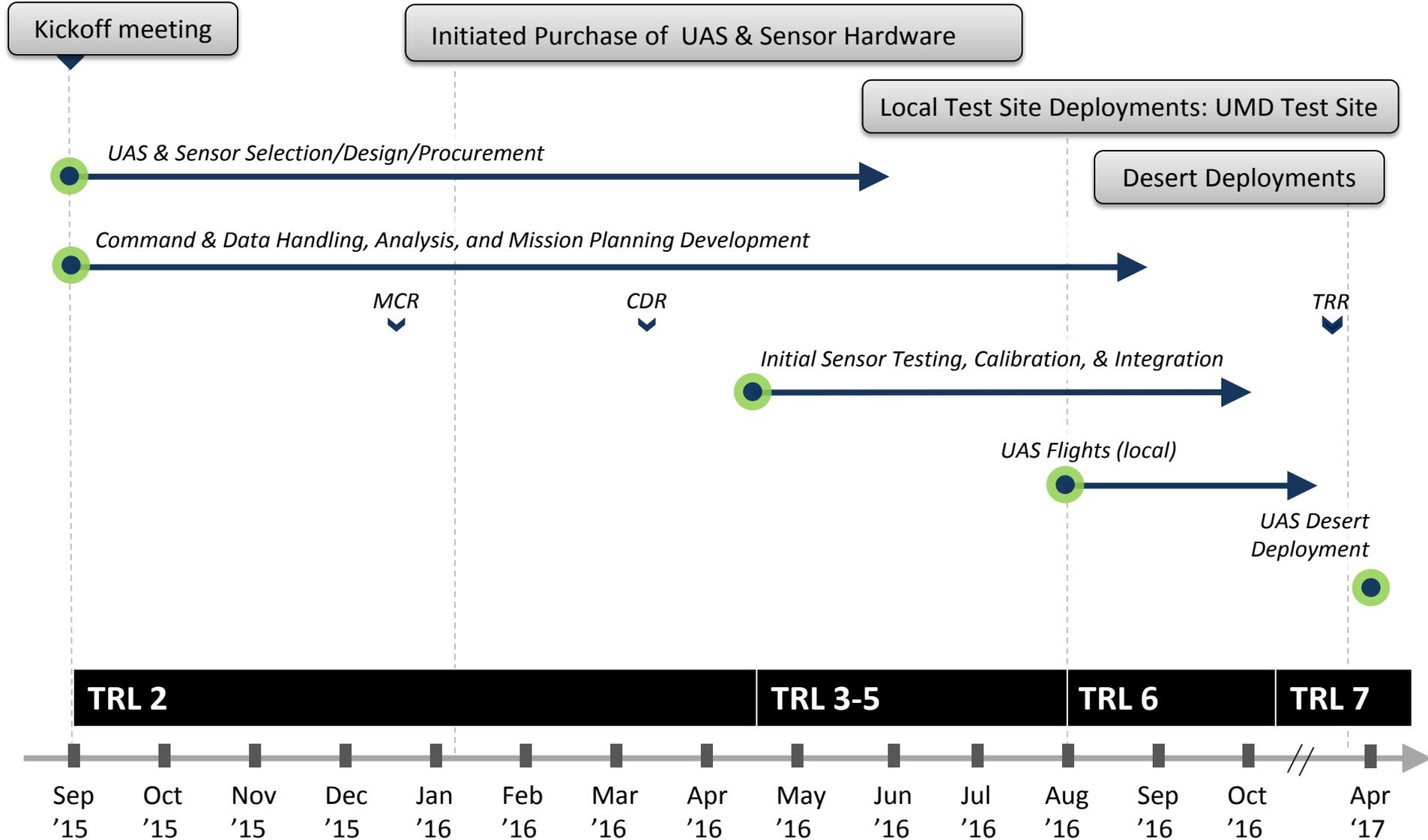


Phase 2: Capability & CONOPS Optimization

Current Phase → Phase 3: Field Campaign Deployment

GOES-R UAS Feasibility Demonstration Study Milestones & Schedule

Drafted & Submitted an initial set of near surface UAS science requirements to the NOAA Unmanned Aircraft Systems Program in January of 2015



GOES-R Near Surface UAS Capability Priorities

1) Hyperspectral (0.35 – 2.5 μm) Reflective Solar Band (RSB) measurements are of highest priority

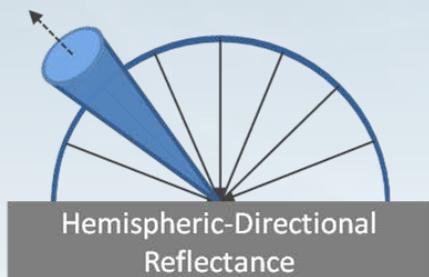
- **Downward Observation (surface)**

Ability to autonomously control the view geometry of the sensor payload(s) for oblique angle data collection of a fixed earth target: Range: 0° (nadir) to 90° (horizon) with a step size of 1° or less

» Near surface ~10 m above ground level (i.e. assume atmosphere is negligible)

Hemispheric-Directional: geometry specified by a cone and a hemisphere

Intended Measurement (Goniometric)



Hemispheric incoming (incoming directional component lost) & directional outgoing geometry



2) Broadband IR (8 – 14 μm) measurements

- Directional Surface Observations (ideally filtered to match ABI spectral bands, primary focus ABI Bands 14-15)

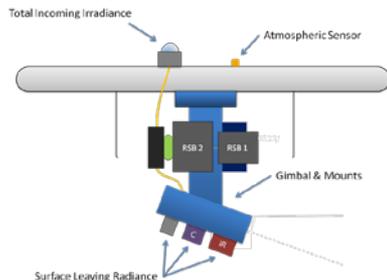
2) High resolution georeferenced imagery (NADIR & Oblique)

- Context imagery of calibration/validation targets & Digital Elevation Model (DEM) generation

Common Requirements for Both Systems

- All sensor measurements have documented SI traceable paths
- All sensor measurement uncertainties are documented and reviewed
- System design shall be flexible to integrate on multiple UAS
- UAS capable of autonomous flight through pre-programmed flight planning
- Meta data to be collected & stored (image acquisition times, sensor look angles, GPS data)

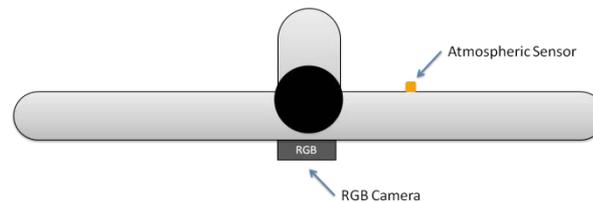
Baseline: Near Surface UAS Systems & Products



Baseline Capabilities:

- Observations over extended regions matching satellite view geometry
- Goniometric observations over a given target (directional hemispheric)

RSB 1 = Compact Hyperspectral (VNIR) Spectrometer
 RSB 2 = Compact Hyperspectral (SWIR) Spectrometer
 IR = LWIR Radiometer(s)
 C = RGB HD Video (Context Imager)

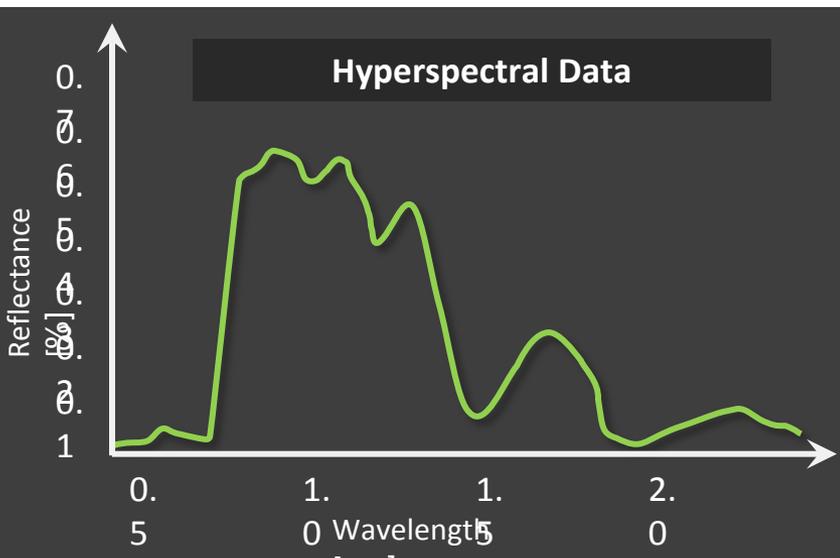


Primary Payload: RSB Hyperspectral (0.35 to 2.5 μm)

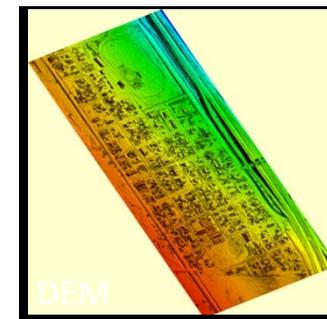
- non-imaging
- Filtered IR radiometers
 Atmospheric profile (near surface)

Primary System

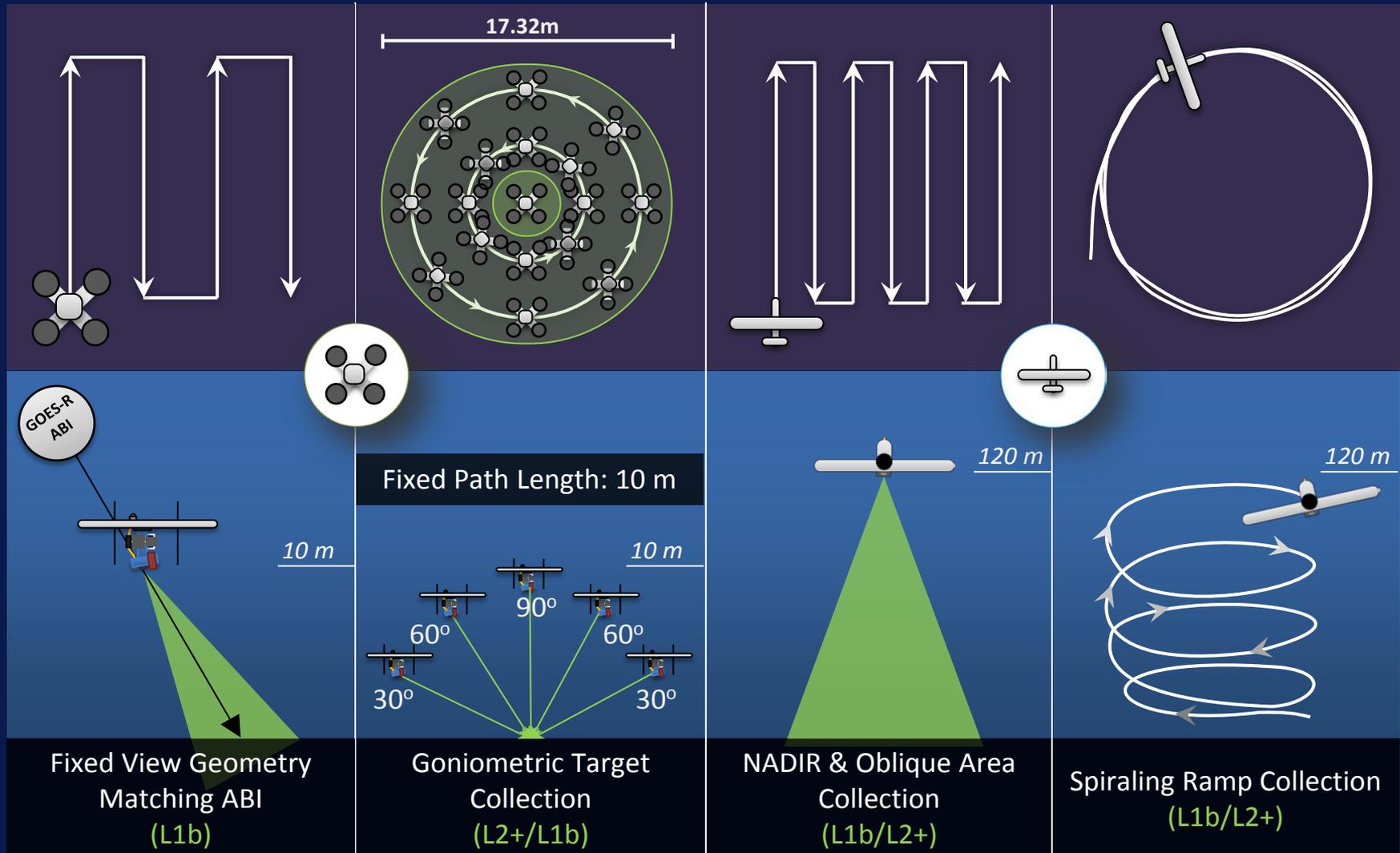
- 2D high resolution georeferenced and orthorectified mosaics
- Digital Elevation Model ($\pm 1-5$ m)
- Atmospheric profiles to maximum collection alt. (~400 ft or 121.9 m)



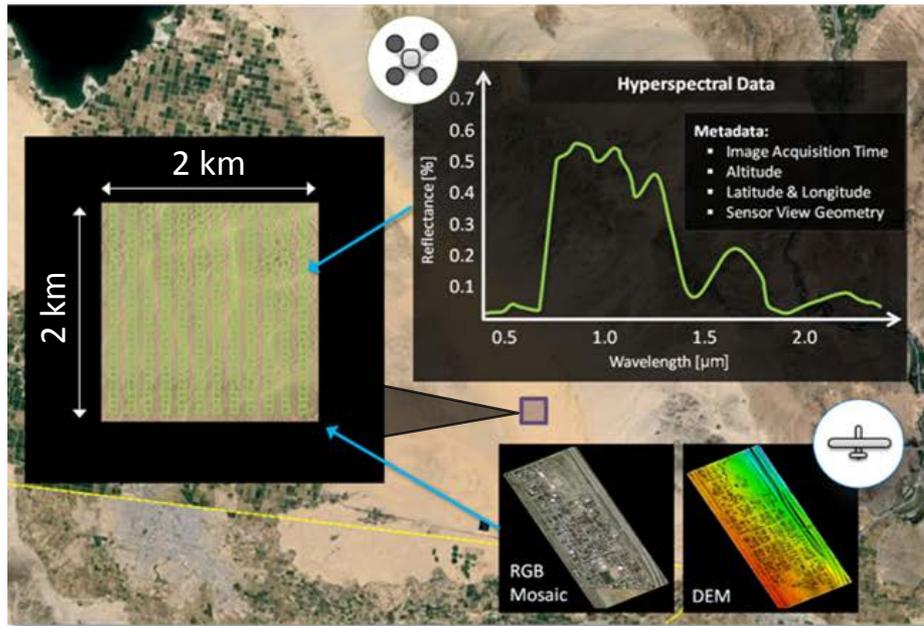
Pix4D Software



Four UAS Baseline Collection Types



Near Surface UAS Measurements Provide Improved Validation Capabilities: Validation of Satellite Data



UAS Capability Can Enhance GOES-R ABI Post-launch Validation Capabilities:

- Provides a pathway to validate radiometric performance post-launch (Reflective Solar Bands & Thermal Emissive Bands surface channels) and product performance uncertainties
- UAS deployments can support long-term monitoring of satellite sensor performance
- Enduring capability for Cal/Val scientist:
 - Near surface UAS campaigns can be replicated numerous times throughout the year at significantly reduced costs in comparison to heritage approaches
 - UAS deployments can support characterization of the degree of uniformity within the given satellite footprint (Ideally, for all reference Cal/Val sites (i.e. fixed ground instruments) in different seasons
- Goniometric surface measurements can be used to check components of model values used in retrieval algorithms



UNIVERSITY OF MARYLAND UAS TEST SITE

A. JAMES CLARK SCHOOL of ENGINEERING



35+ UAS (fixed & rotor wing, 3 lbs to 160 lbs)

- 1 x UAVS Talon 240G
- 3 x UAVS Talon 120LE
- 3 x AV Wasp
- 3 x AV Raven
- 3 x AV Dragon Eye
- 1 x "Dragon Pi"
- 3 x Apprentice S15e
- 1 x FireFLY 6 hybrid
- 1 x DJI S-1000
- 2 x DJI S-900
- 1 x UAVS Phoenix 60
- 1 x UAVS Phoenix ACE LE
- 1 x UAVS Phoenix ACE XL
- 6 x 3DR Iris
- 1 x 3DR Solo
- 5 x DJI Phantom 3
- 1 x DJI Inspire

Payloads/Sensors specific to mission requirements

- Established Aug 2014 as one of six FAA UAS Test Sites
 - Partnered with Virginia & New Jersey
 - Located in Southern MD near Patuxent River NAS (Navy UAS test & eval)
- Govt/Academia/Industry research customers – focus on integrating UAS into National Airspace System & civil/commercial applications of UAS
- Robust airworthiness process; reachback to expertise at College Park
- Flight ops under public COAs, FAA Part 107, or foreign/international rules; airspace access nationwide including segregated airspace
- Major Research Projects
 - FAA, DHS: Airspace Intrusion Detection (1st *legal* UAS flight in Class C)
 - 1st civil UAS cargo flight across Chesapeake Bay (simulated medical supplies)
 - NOAA: GOES-R satellite cal/val & NERR mapping
 - GWU, USNA: Ship Air Wake Analysis (flown from YP boat in Chesapeake Bay)
 - US Navy: Open source autopilot (analysis of alternatives)
 - NASA: UAS Traffic Management (air traffic control for UAS)
 - Public Safety Agencies: life preserver drop, missing person search, comms relay, accident scene reconstruction, emergency vehicle support, radiation detection
 - Agriculture/Aquaculture/Anthropology/Geology: Aerial Surveys & 3D Mapping
 - AQWUA: A Quad with Underwater Abilities (fly/swim)
 - BVLOS/BLOS Ops (current requirement is visual LOS only)
 - Collaborative Control (UAS swarming)



Prototype Fixed-Wing System: UAS + Payloads

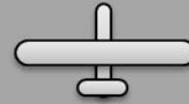
1. **High Resolution Camera:**
 - High resolution RGB camera
2. **Atmospheric Sensor:**
 - T, RH, & Px profiles

Baseline Capabilities:

- 2D high resolution georeferenced and orthorectified mosaics (NADIR & Oblique)
- Digital Surface Model ($\pm 1-5$ m)
- Atmospheric profiles to maximum collection alt. (~400 ft or 121.9 m)



Secondary System (TRL 9)

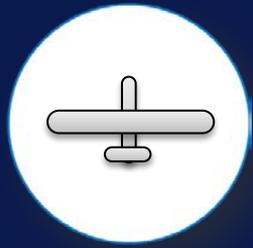


Fixed-Wing UAS



Talon120 Specifications

Length: 6' Wingspan: 12.5'
Endurance: 2.0 -2.5 hours
Range: 8 mile LOS
Fully autonomous system
Payload Capacity: 2.5 lbs



Fixed-Wing UAS Payloads



baseline

Cannon S100
(RGB)

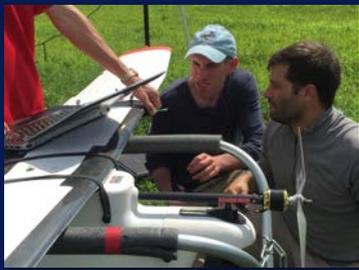
Imaging

Cannon
TRL 9

Atmospheric Sensor
(T, Px, RH)

Non-Imaging

InterMet Systems
TRL 7



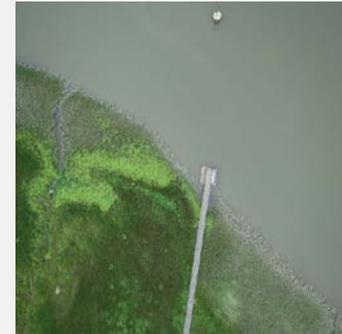
GOES-R UAS Feasibility Demonstration Study: Successful Fixed-Wing UAS functional & operational performance demonstrations



Completed successful test flights at the:

- University of Maryland (UMD) UAS test site in Bushwood, MD on August 3, 2016
- NOAA National Estuary Research Reserve (NERR) in Jug Bay, MD on August 8, 2016 – UAS test data provided to NOAA NERR as operational data
- Resulting products: 2D & 3D geo-referenced maps

NADIR



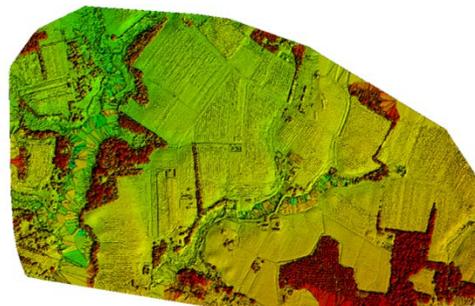
Oblique
Imagery



2D Geo-Referenced
Orthomosaic



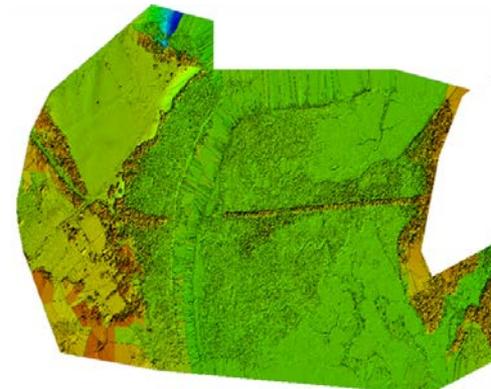
3D Digital Surface
Model



2D Geo-Referenced
Orthomosaic



3D Digital Surface



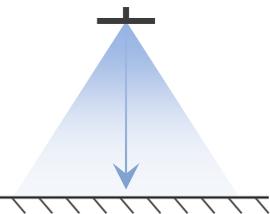
Fixed-Wing UAS Sample data

2D Georeferenced Imagery Mosaic - Flight Altitude of 700 ft



Fixed-Wing UAS Collection (NADIR to 45°)

NADIR Collection:



North

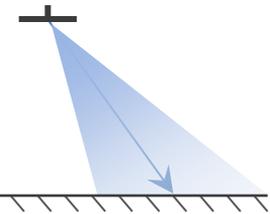
West

East

South



Oblique Collection (~45°)



2D Georeferenced Orthomosaics Products NADIR + Oblique Imagery vs Nadir Only Imagery

NADIR + Oblique



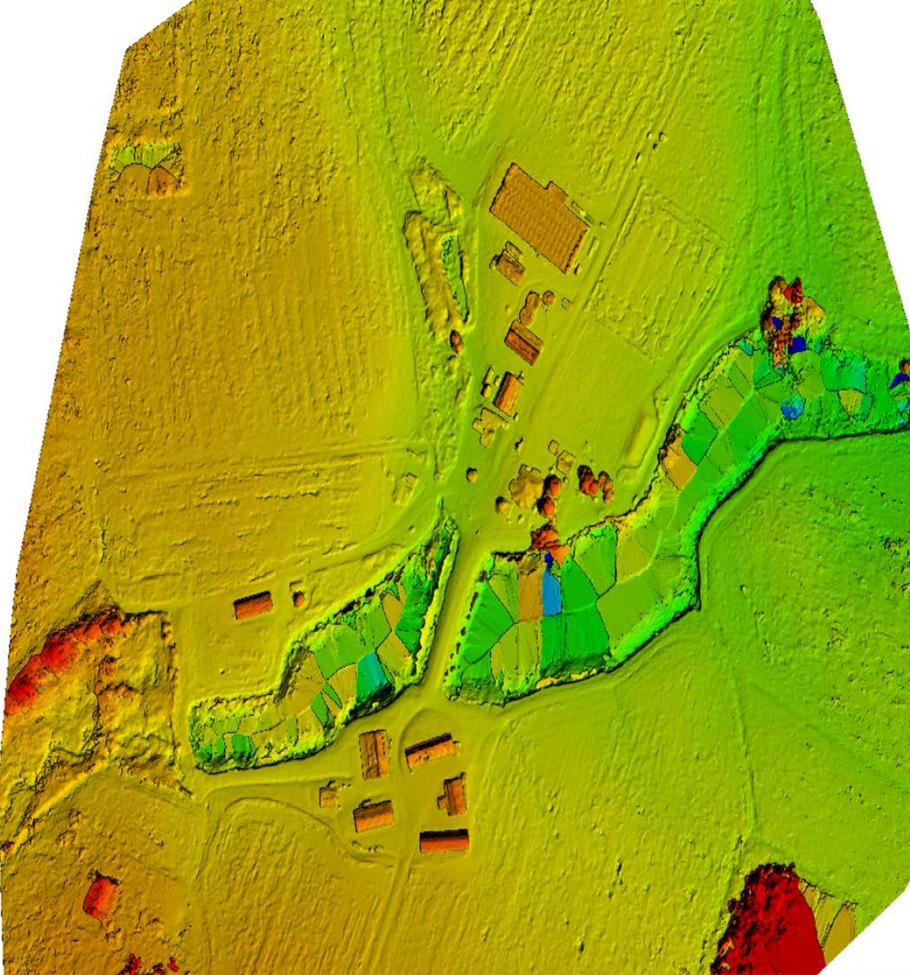
NADIR Only



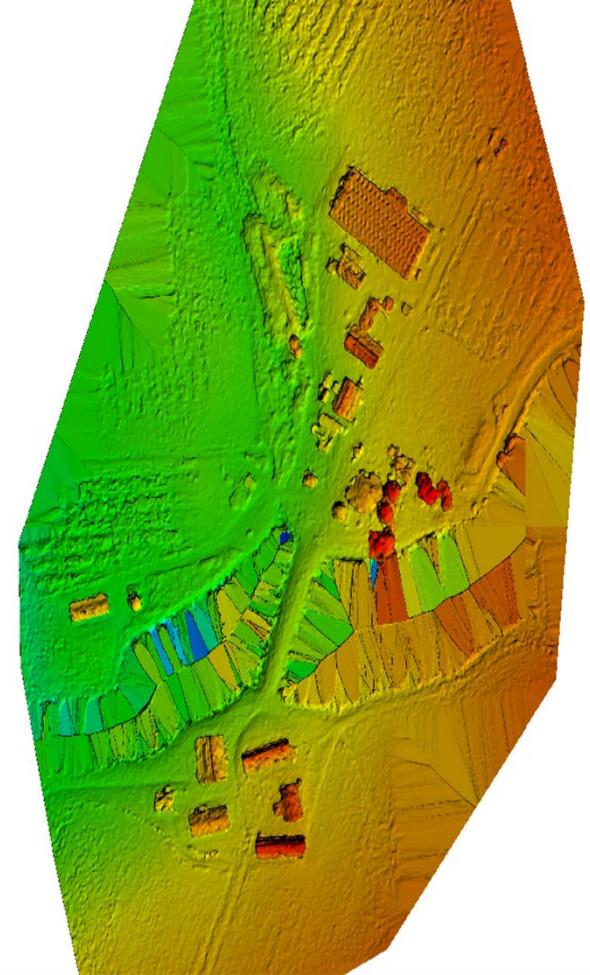
3D Georeferenced Digital Surface Model (DSM) Products

NADIR + Oblique Imagery & Nadir Only Imagery

NADIR + Oblique



NADIR Only



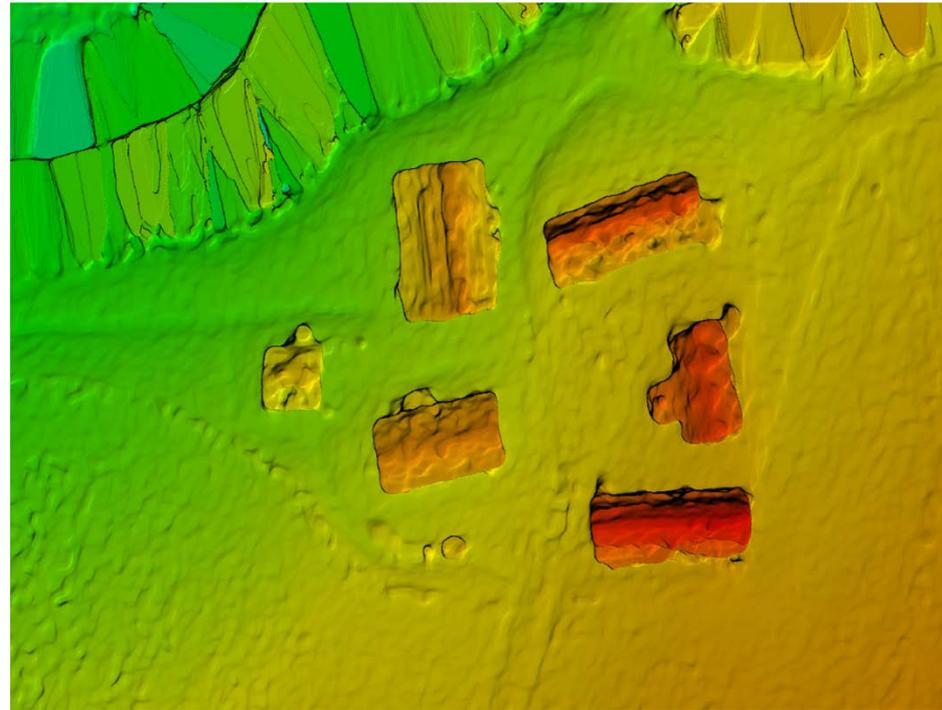
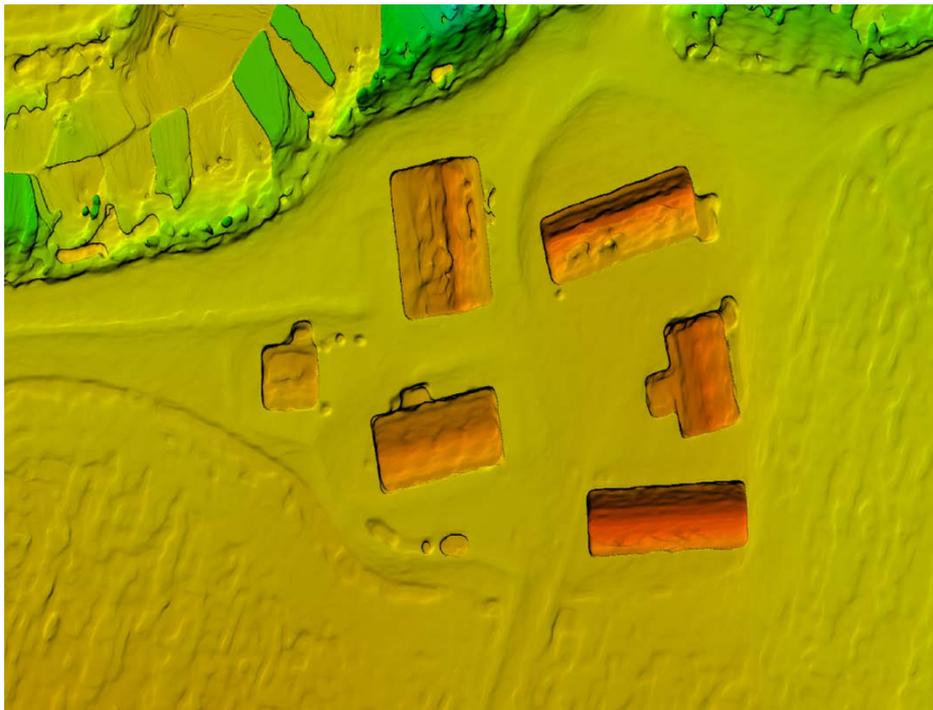
3D Georeferenced Digital Surface Model (DSM) Products

NADIR + Oblique Imagery & Nadir Only Imagery

-- Zoom In --

NADIR + Oblique

NADIR Only



- NADIR + Oblique imagery dataset produced an enhanced DSM (better defined structure) vs the NADIR only dataset

3D Georeferenced Digital Surface Model (DSM) Products NADIR + Oblique Imagery & Nadir Only Imagery -- Zoom In --

NADIR Only



3D Georeferenced Digital Surface Model (DSM) Products NADIR + Oblique Imagery & Nadir Only Imagery -- Zoom In --

NADIR + Oblique



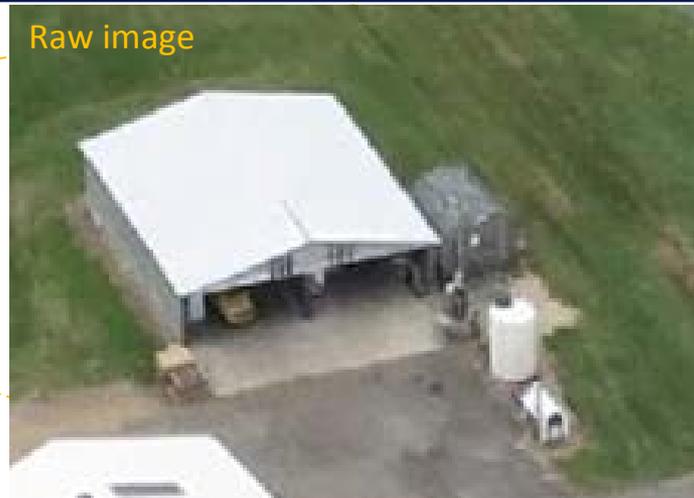
3D Georeferenced Digital Surface Model (DSM) Products NADIR + Oblique Imagery & Nadir Only Imagery

-- Zoom In --

NADIR + Oblique



Raw image



NADIR Only

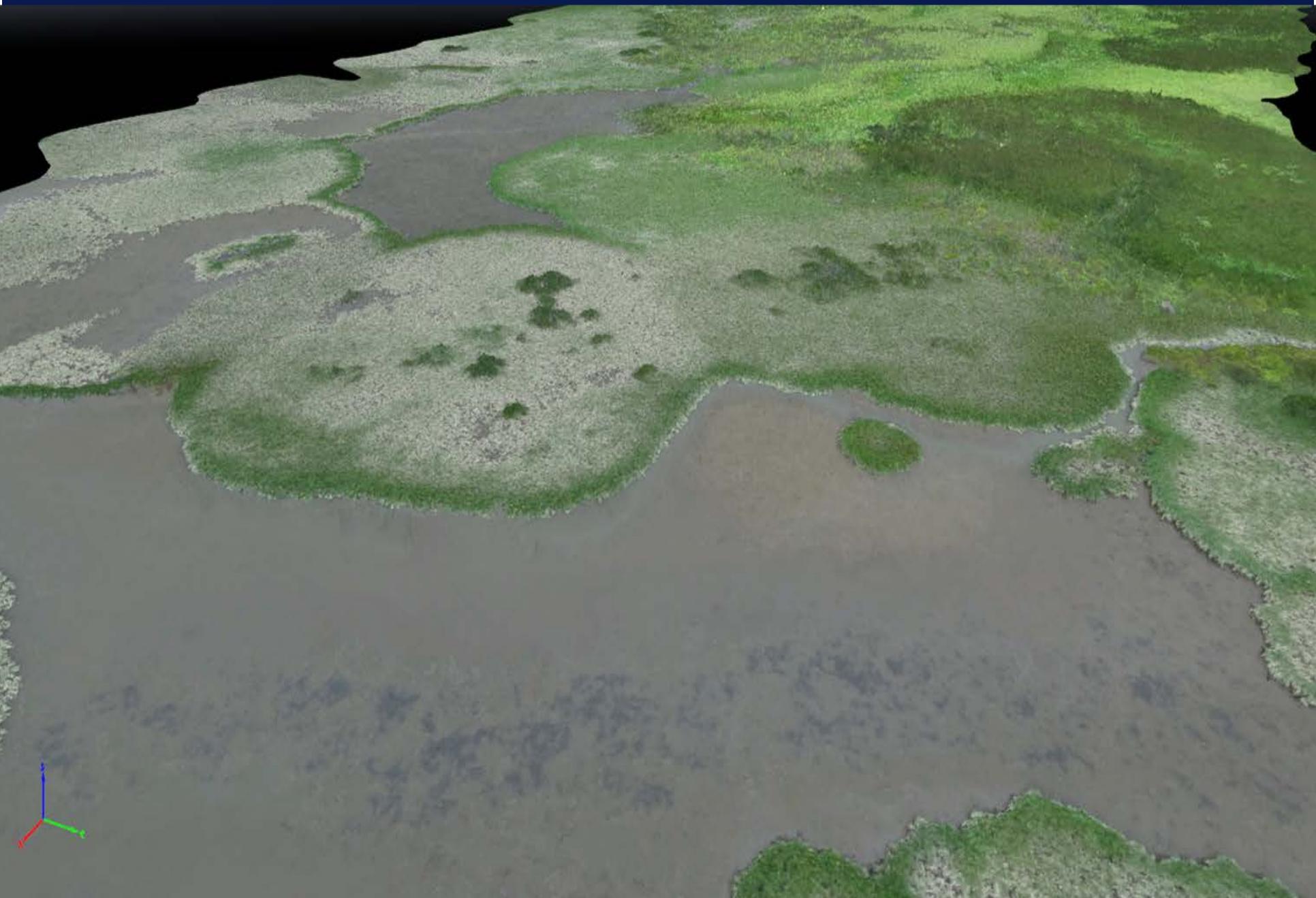


2D Geo-Referenced Orthomosaic





NOAA National Estuary Research Reserve (NERR) Jug Bay, MD – August 8, 2016

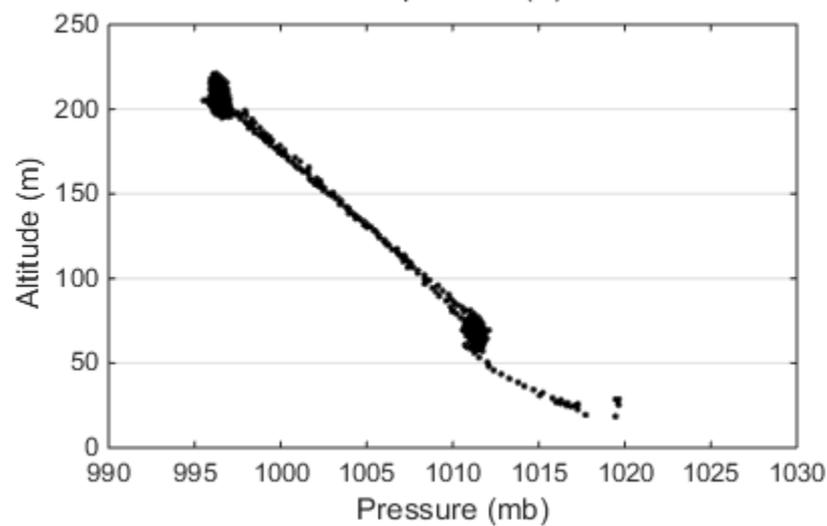
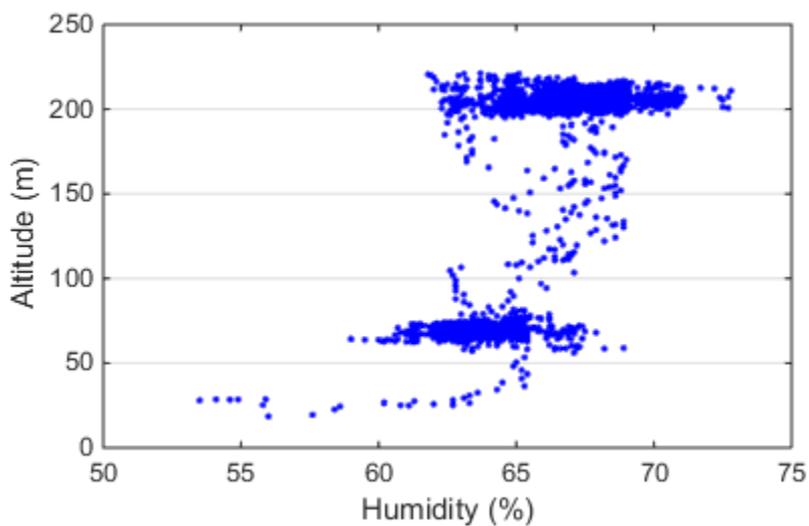
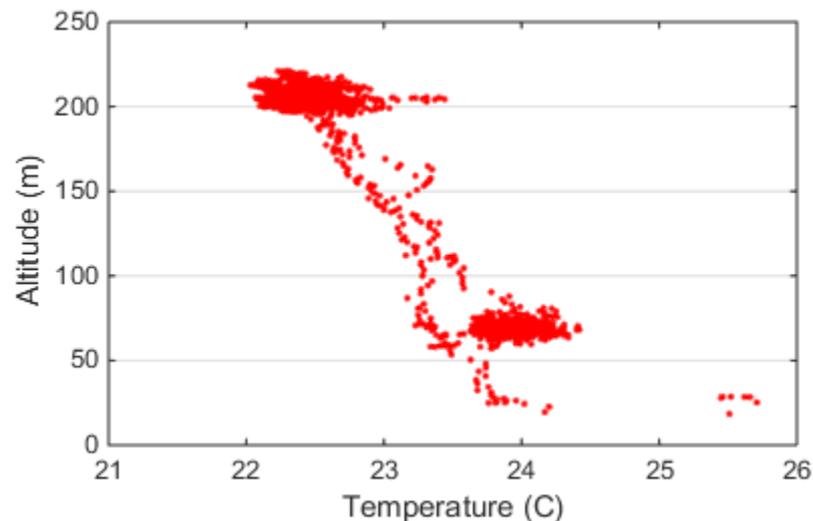




Atmospheric Sensor: Functional Performance Testing



University of Maryland (UMD) UAS test site in Bushwood, MD on August 3, 2016



Prototype Rotary System: UAS + Payloads

1. Reflective Solar Band (RSB) Sensor Suite:

- Hyperspectral coverage from 0.35 to 2.5 μm
 - Downward (directional)

2. IR Radiometer:

- Broadband IR – 8-14 μm /potentially filtered to match the ABI channels

3. Context Imager:

- RGB HD video - context imager

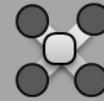
4. Atmospheric Sensor:

- T, RH, and Px profiles

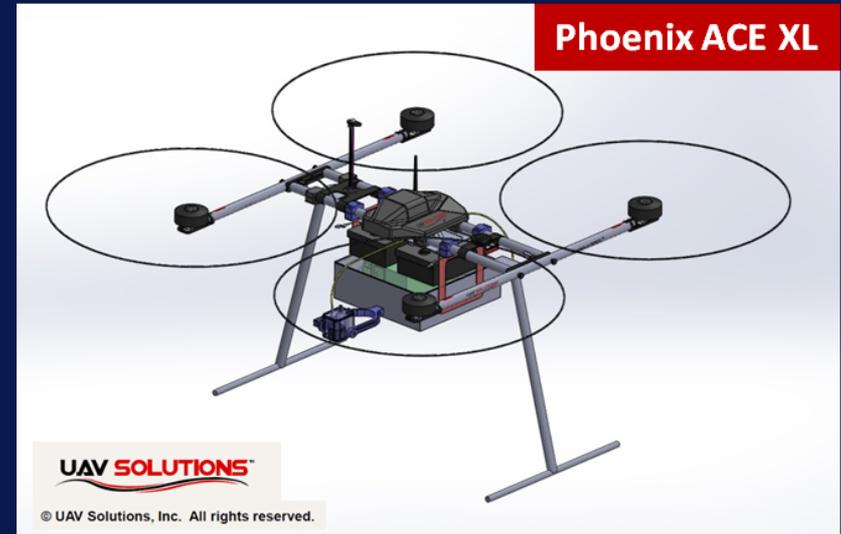
Baseline Capabilities:

- Observations over extended regions matching ABI view geometry
- Goniometric observations over a given target (directional hemispheric)

Primary System (TRL 7)



Rotary UAS

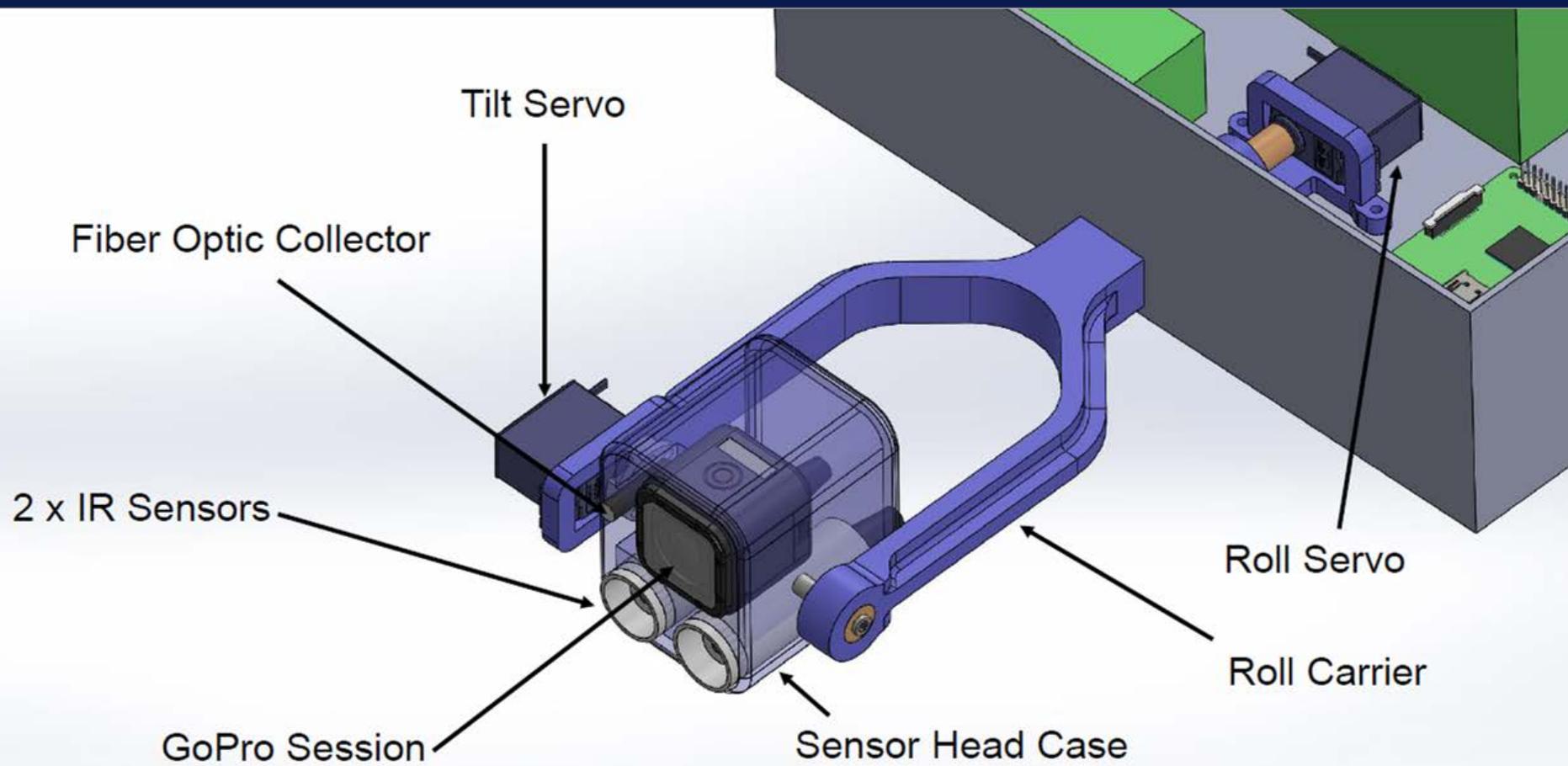


Phoenix ACE XL Specifications

Endurance: 30 minutes of collection
Fully autonomous system
Payload Capacity: 10-12 lbs

Customized electronic enclosure and autonomous 2 axis gimbal

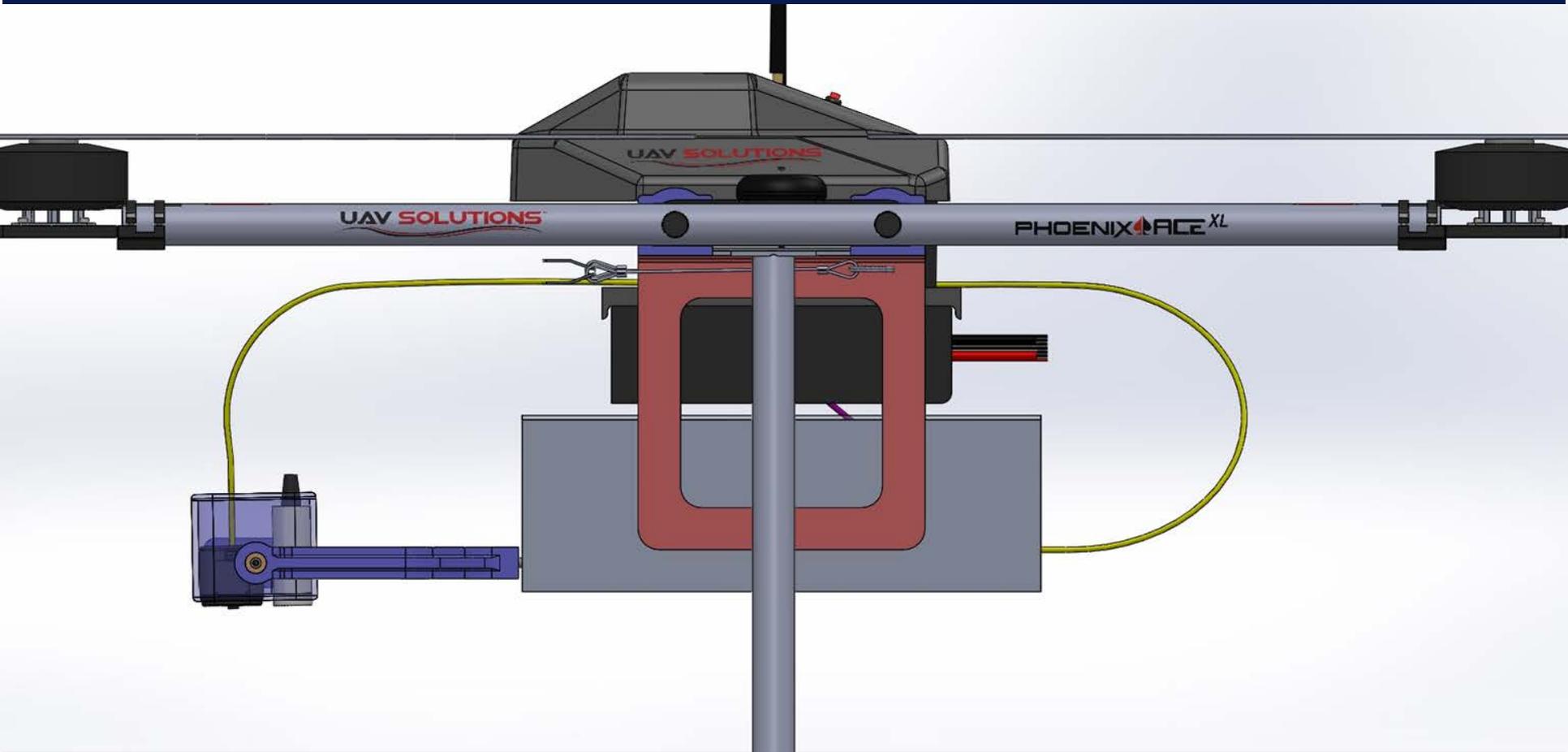
GOES-R Prototype Rotary UAS: Downward Observations



GOES-R Prototype Rotary UAS:

Gimbal & Fiber Motion

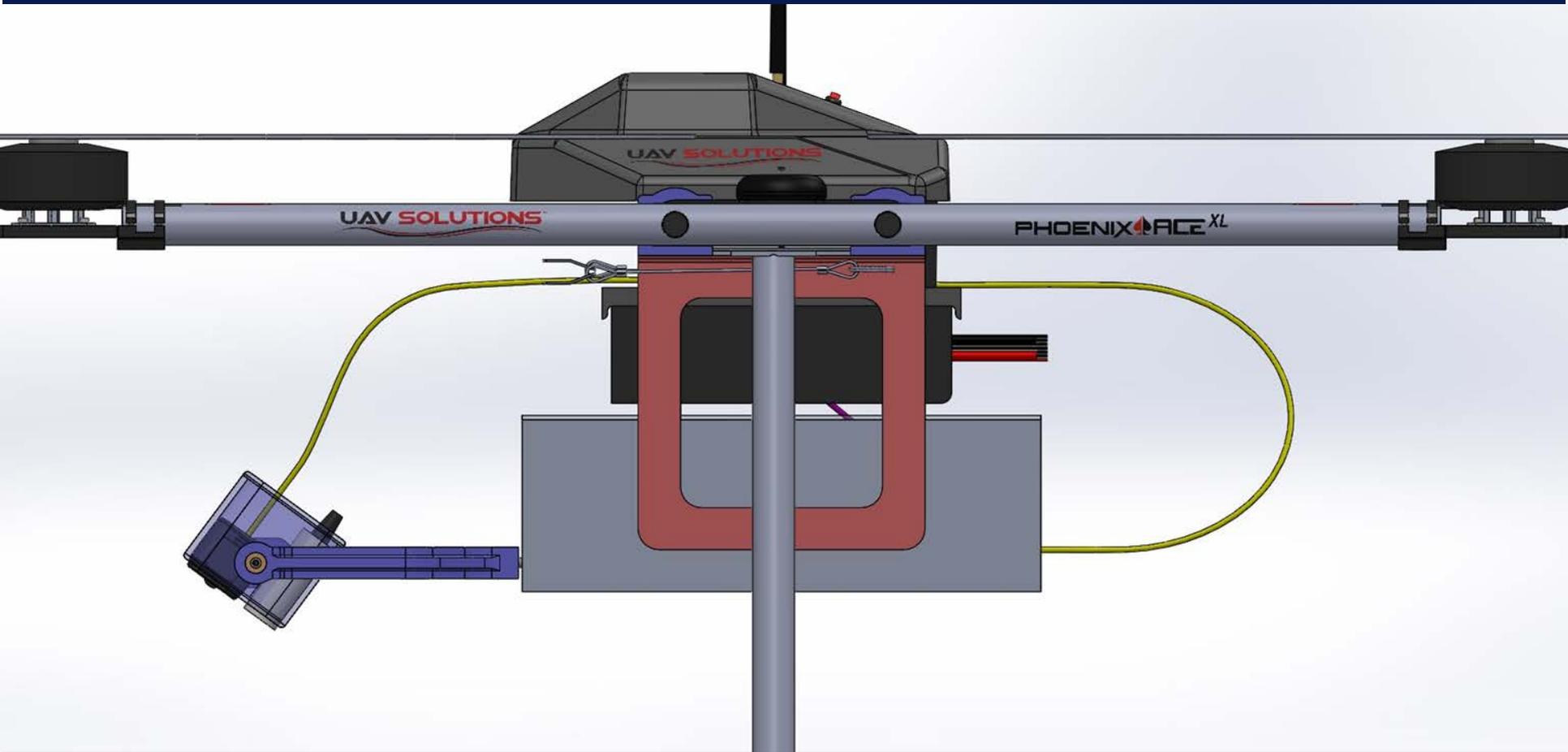
0° NADIR Viewing

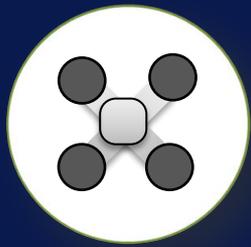


GOES-R Prototype Rotary UAS:

Gimbal & Fiber Motion

45° NADIR Viewing





Rotary UAS Payloads



RSB Sensor Suite

VNIR Spectrometer
(0.35 – 1.1 μm)

Non-Imaging

Ocean Optics
TRL 9



SWIR Spectrometer
(0.9 – 2.6 μm)

Non-Imaging

ARCoptix
TRL 7



IR Radiometer

Broadband IR
(8.0 – 14 μm)

Non-Imaging

Apogee Instruments
TRL 9



Context Imager

Context HD Video
(RGB)

Imaging

GoPro
TRL 9



Atmospheric Sensor

Atmospheric Sensor
(T, Px, RH)

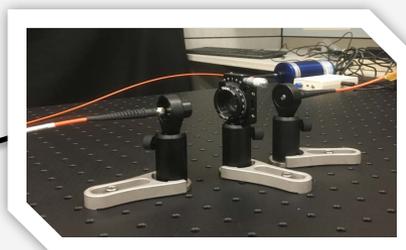
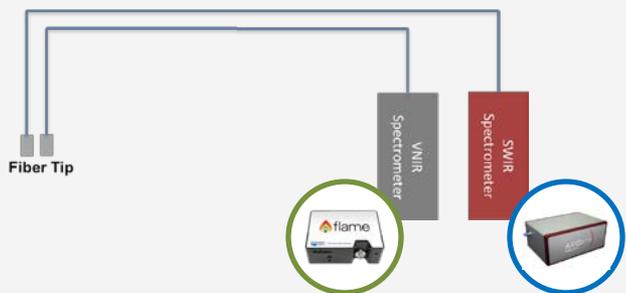
Non-Imaging

InterMet Systems
TRL 7

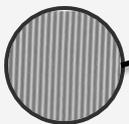
Radiometric/Geometric Calibration & Characterization

NOAA Calibration Center laboratory developed for UAS payload calibration & characterization to ensure data quality

Reflective Solar Band Sensor Suite



Polarization Sensitivity



Calibration



Wavelength

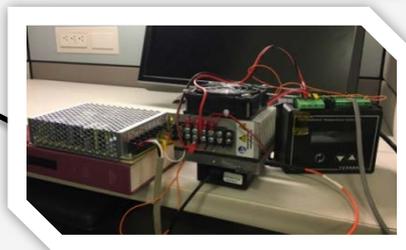
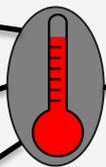


Temperature Effects

Dark counts



Responsivity



[Pearlman et al. SPIE 2016]

Camera Calibration



To enable overlaying of sensor footprints in context imagery

Distorted



Corrected

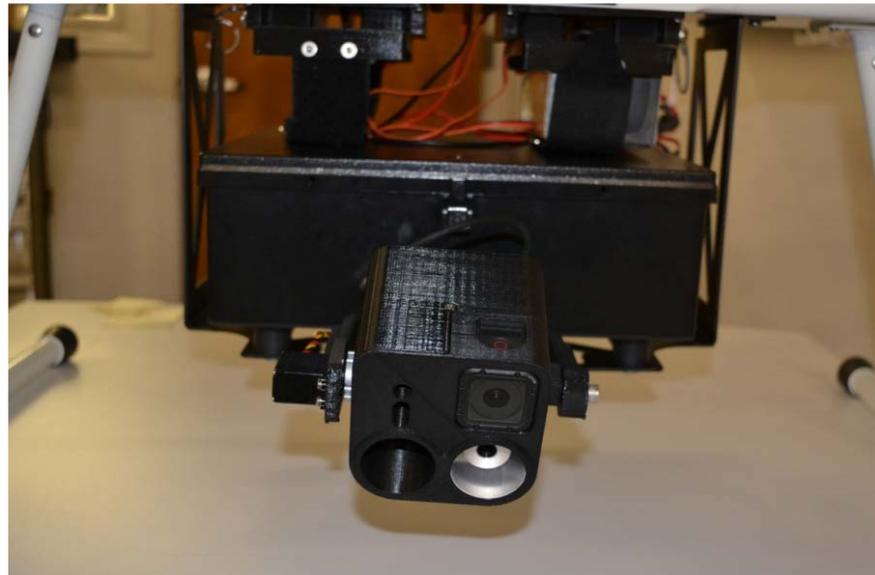


[www.OpenCV.org]

Prototype Rotary UAS Developed



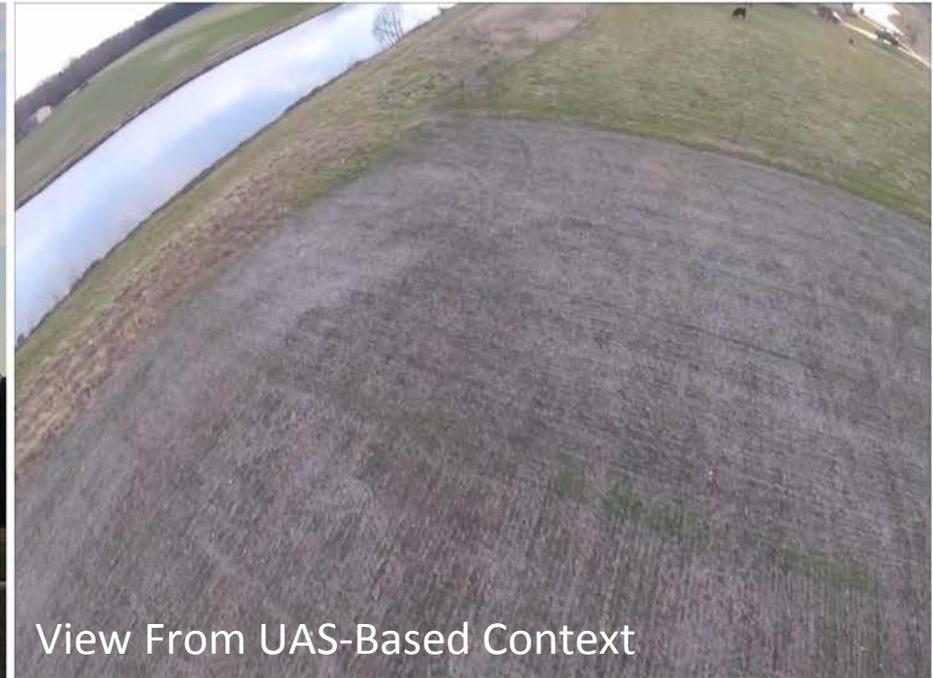
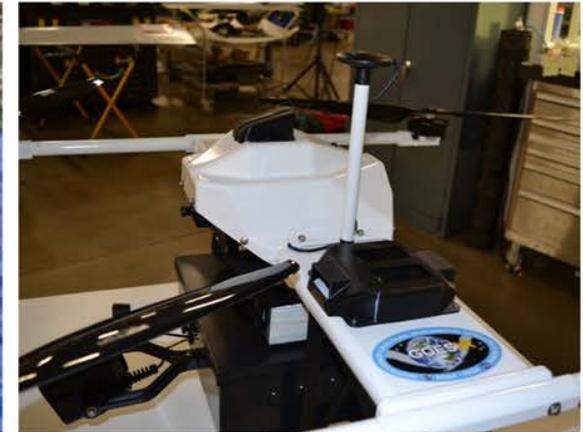
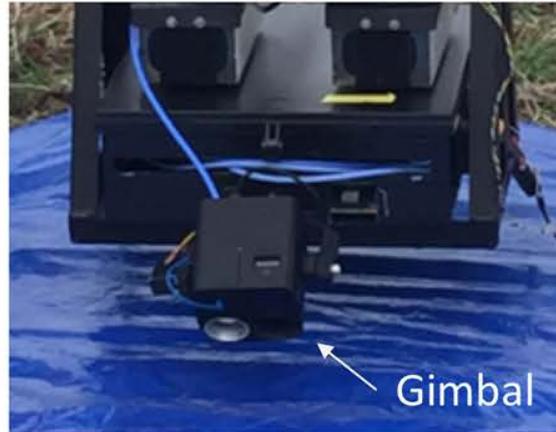
Flight Testing Without Payloads



Flight Testing With Payloads

Prototype Rotary UAS Developed (TRL 7):

- Autonomous Mission Planning
- 2-axis stabilized gimbal (pointing capability)
- Hyperspectral VNIR/SWIR & thermal IR broadband sensors
- Context imager



View From UAS-Based Context Imager

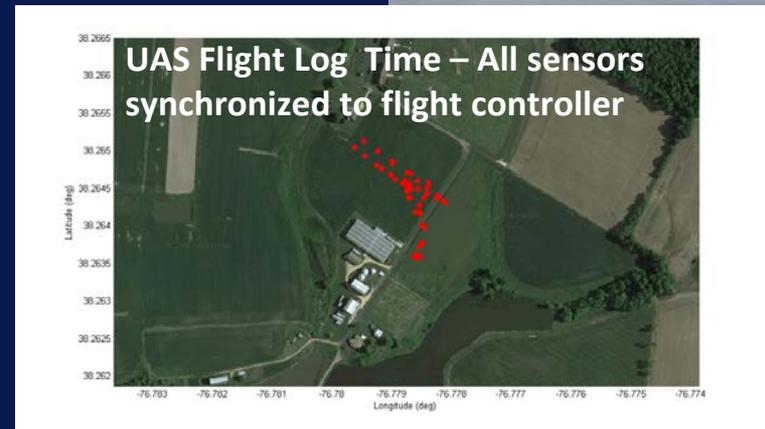


GOES-R UAS Feasibility Demonstration Study: Successful Rotary UAS functional performance demonstrations



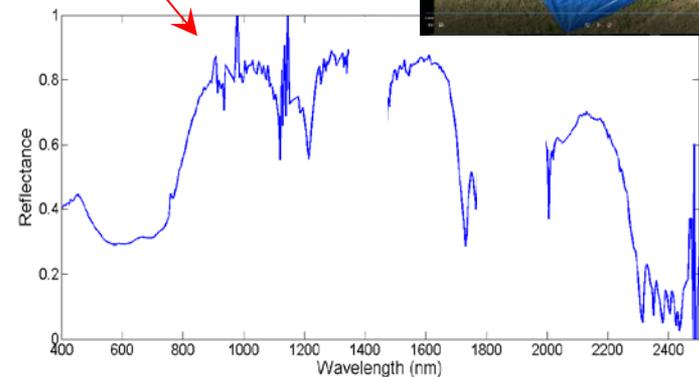
Completed successful functional test flights:

- NOAA National Estuary Research Reserve (NERR) in Jug Bay, MD on November 11, 2016
- University of Maryland (UMD) UAS test site in Bushwood, MD on January 13, 2017
- Demonstrated functional performance of prototype UAS & payloads:
 - Hyperspectral & thermal IR geo-referenced products



Blue Tarp:

- 1) In-Flight Photo
- 2) Hyperspectral Spectra



Operational Environment Testing: GOES-R Field Campaign April 3-7, 2017

GOES-R near surface UAS feasibility demonstration study concludes with an operational environmental test in Red Lake, AZ during the GOES-R field campaign:

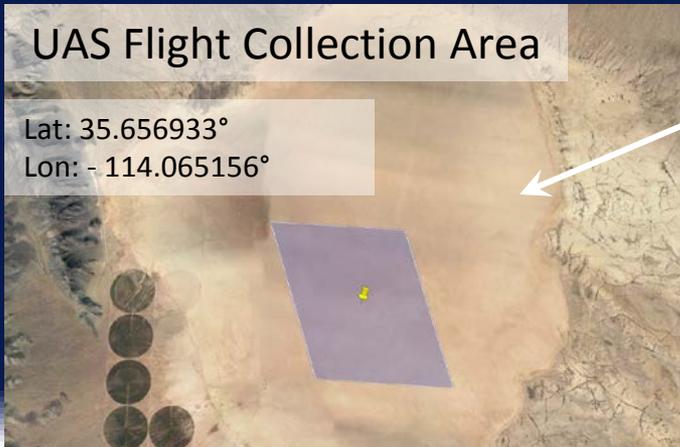
- Test data will provide reference data in support of GOES-R ABI post-launch validation efforts

Reference Measurements to be collected:

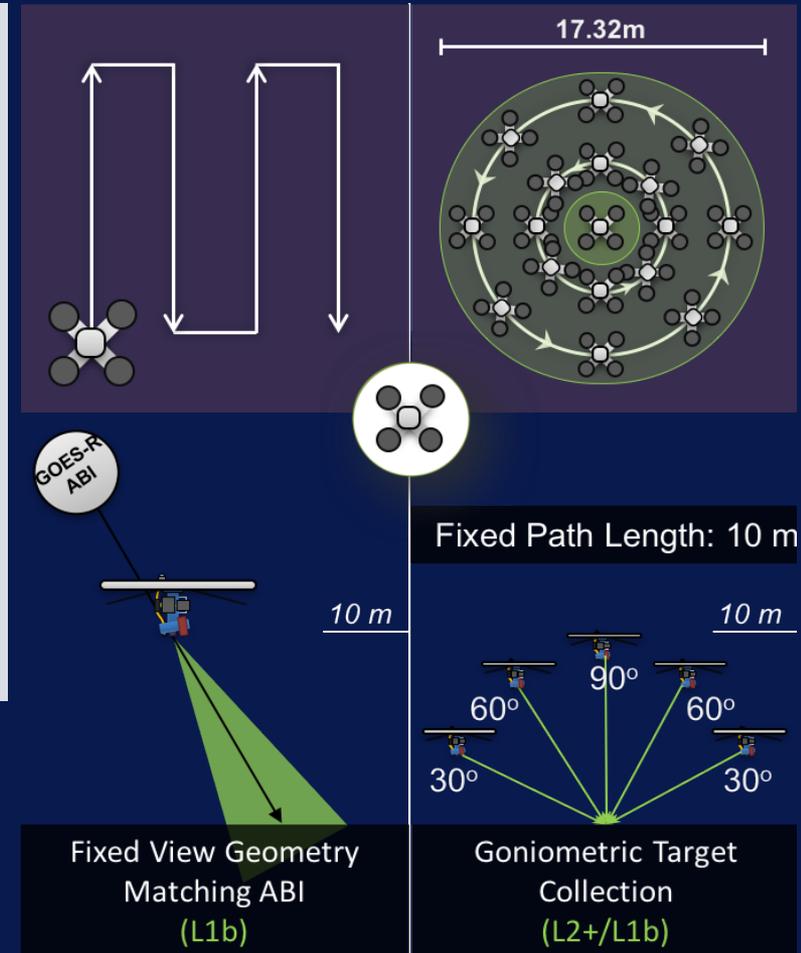
- UAS rotary geo-referenced products:
 - » Hyperspectral surface reflectance
 - » Surface effective temperature
- Mobile SURFRAD station:
 - » Automated surface & atmospheric measurements

UAS Flight Collection Area

Lat: 35.656933°
Lon: - 114.065156°



Red Lake, AZ



Pre-Deployment Site Survey Completed
February 7-8, 2017

Challenges Addressed

- **Open Source Flight Controller**
 - Enables access to the UAS metadata (Pixhawk)
- **Autonomously Controlled Payload Gimbal**
 - Customized 2-axis servo to ensure the UAS based sensors match the view geometry of the satellite sensor
 - Mission Planner & flight controller (Pixhawk) – capability improvements identified
- **Payload Command and Data Handling**
 - Developed a multi-sensor architecture using two single board computers (Raspberry PI) in parallel
- **Time Synchronization of the UAS System**
 - Ensures proper metadata tagging between sensor payloads and UAS flight controller and improved sensor product navigation
- **Near Surface Flight Operations: ~10 m**
 - Radiative transfer simulations validated the approach
- **Oblique Image Capability Added**
 - Enables enhanced image analysis & 3D geo-referenced imagery products via structure from motion techniques



Summary



- Developed & demonstrated a rotary & a fixed-wing UAS capabilities in support of advanced capability development efforts of the GOES-R field campaign:
 - Geo-referenced hyperspectral and thermal IR measurements
 - 2D & 3D geo-referenced products over a targeted area
 - GOES-R feasibility demonstration study completes with an operational environmental test (rotary UAS) in Red Lake, AZ (April 3-7, 2017) during the GOES-R field campaign and will provide reference data in support of GOES-R ABI post-launch validation
- Rotary UAS – TRL 7 (current status)
- Fixed-Wing UAS – TRL 9
- Developing web-based data discovery and visualization tools to enhance data sharing and analysis
- Developing image quality and data product levels to optimize end products
- The GOES-R near surface UAS feasibility demonstration study supports advanced capability development for the GOES-R field campaign.
 - Final report to be completed summer 2017

BACK-UP

GOES-R Advanced Baseline Imager

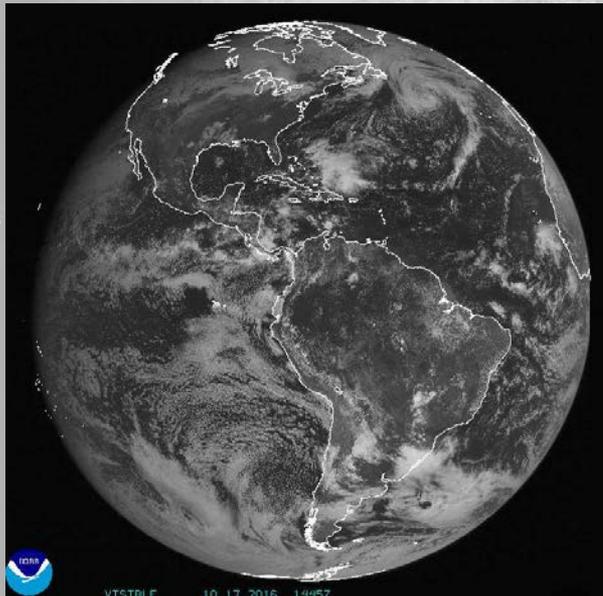
ABI MODES OF OPERATION

- **Full Disk:** Hemispheric Coverage of 83° local zenith angle, temporal resolution of 5-15 minutes, and spatial resolution of 0.5 to 2km
- **Mesoscale:** Provides coverage over a 1000x1000km box with a temporal resolution of 30 seconds, and spatial resolution of 0.5 to 2km.
- **Continental US:** The CONUS scan is performed every 5 minutes, providing coverage of the 5000km (E/W) and 3000km (N/S) rectangle over the United States. The spatial resolution is 0.5 to 2km.
- **Flex Mode:** The flex mode will provide a full disk scan every 15 minutes, a CONUS every 5 minutes, and two mesoscale every 60 seconds (or one sub-region every 30 seconds).

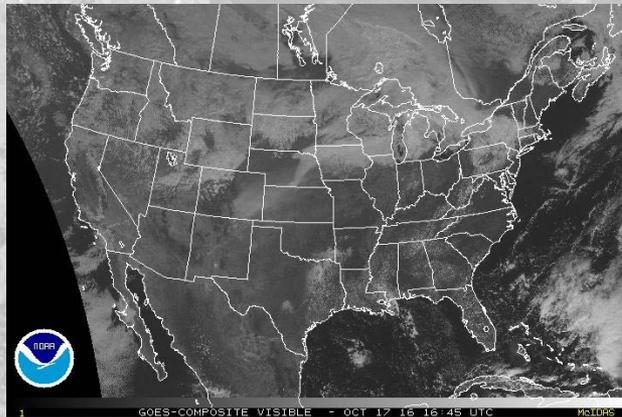
COMPARISON GOES-R SERIES ABI VS CURRENT GOES

ATTRIBUTE :	ABI	CURRENT GOES IMAGER
Spectral Coverage	16 bands	5 bands
Spatial Resolution		
0.64 μm Visible	0.5 km	~ 1 km
Other visible/near-IR	1.0 km	n/a
Bands ($>2 \mu\text{m}$)	2 km	~ 4 km
Spatial Coverage		
Full Disk	4 per hour	Scheduled (3 hrly)
CONUS	12 per hour	n/a
Mesoscale	30 or 60 sec	~4 per hour
		n/a
Visible (reflective bands)		
On-orbit calibration	Yes	No

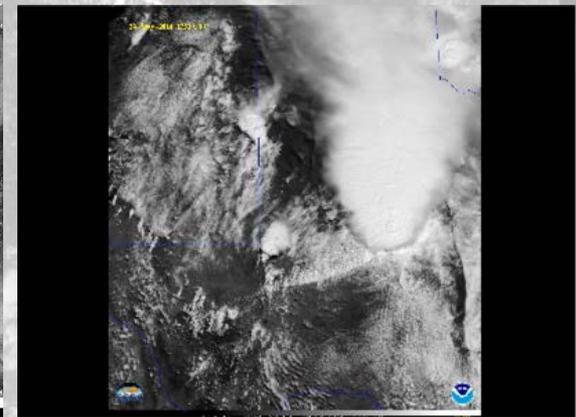
Full Disk



CONUS



Mesoscale



ABI Field Campaign Approach:

Primary Objective: provide validation of ABI L1b spectral radiance observations to validate SI traceability

Secondary objective: provide surface and atmospheric geo-physical measurements to support L1b & L2+ product validation

Targets of Interest:

Desert Open Ocean



Vegetation

Clouds

